Report Produced By:	Charles King	Address:		P	hone:	
USARL Research Date:	Airworthiness	Directive	Compliance Record		File ID:	NEW
<u>M</u> anufacturer Textron Lycoming	<u>M</u> odel IO-540-G1D5				77.77	Not Specified Not Specified
AD Number Effective date	Description	Complied Date Time	Status  Method of Compliance/Applicability	Once or Recur	Next Due Date Time	1. Facility 3. Cert. Num. 2. Cert. Type 4. Author. by
63-14-03 07/06/1963	OIL PUMP DRIVE SHAFT			Recur		1. 2. 3. 4.
©ATP	©ATP			©ATP	Signature:	
66-20-04 08/27/1966	TO PREVENT FURTHER FAILURES OF OIL FILTER ADAPTER GASKET, P/N 74904			Once		1. 2. 3. 4.
©ATP	©ATP			©ATP	Signature:	
73-23-01 R(4) 01/13/1977	TO PREVENT PISTON PIN FAILURES RESULTING FROM GRINDING CRACKS WHICH OCCURRED D  ©ATP			Once	Signature	1. 2. 3. 4.
75-08-09 R(3) 08/18/1977	TO PREVENT OIL PUMP FAILURES, INSPECT, REPLACE AND ASSEMBLE THE OIL PUMP DRIVE			Once	Signature	1. 2. 3. 4.
75-09-15 04/30/1975	TO PREVENT POSSIBLE FUEL STARVATION TO THE ENGINE			Once		1. 2. 3. 4.
©ATP	©ATP			©AT	0.5	re: 1.
78-23-10 11/07/1978	TO PREVENT AN IN-FLIGHT POWER LOSS DUE TO AN OVER RICH CONDITION, CONTD.				TP Signatu	2. 3. 4.
©ATP	©ATP					

Report Produced By:	Charles King	Address:		F	Phone:	
JSARL Research Date:	Airworthiness 03/03/2004	Directive	Compliance Record	-16/2	File ID:	NEW
<u>Manufacturer</u> Fextron Lycoming	<u>M</u> odel IO-540-G1D5	N'egille				: Not Specified : Not Specified
AD Number Effective date	Description	Complied Date Time	Status Method of Compliance/Applicability	Once or Recur	Next Due Date Time	1. Facility 3. Cert. Num. 2. Cert. Type 4. Author. by
79-04-05 09/26/1979	TO PREVENT AN IN-FLIGHT POWER LOSS DUE TO THE SEPARATION OF THE P/N 2529192 RE			Once		1. 2. 3. 4.
©ATP	©ATP			©ATP	Signature:	
81-18-04 R2 06/07/1982	Superseded by 96-09-10			Once		1. 2. 3. 4.
©ATP	©ATP			©ATP	Signature:	
83-22-04 11/04/1983	TO PREVENT ENGINE FAILURE DUE TO FATIGUE FAILURE OF THE INJECTOR FUEL DIAPHRAG			Once		1. 2. 3. 4.
©ATP	©ATP			©ATP	Signature:	
91-08-07 C 05/06/1991	TO PREVENT ENGINE POWER LOSS AND POSSIBLE LOSS OF THE AIRCRAFT			Once		1. 2. 3. 4.
©ATP	©ATP			©ATP	Signature:	
91-14-22 08/19/1991	TO PREVENT LOOSENING OR FAILURE OF THE CRANKSHAFT GEAR RETAINING BOLT, WHICH M			Recur		1. 2. 3. 4.
©ATP	©ATP			©ATP	Signature:	
92-12-05 07/10/1992	TO PREVENT PISTON PIN FAILURE, OR PISTON RELEASE, AND ENGINE FAILURE			Once		1. 2. 3. 4.
©ATP	©ATP			©ATP	Signature	

Report Produced By:	Charles King	Address:			Phone:	
USARL Research Date:	Airworthiness	Directive	Compliance Record		File ID	NEW
Manufacturer  Textron Lycoming	Model IO-540-G1D5				-	: Not Specified : Not Specified
AD Number Effective date	Description	Complied Date Time	Status  Method of Compliance/Applicability	Once or Recur	Next Due Date Time	1. Facility 3. Cert. Num 2. Cert. Type 4. Author. by
93-02-05 06/14/1993	Superseded by 2002-26-01			Recur		1. 2. 3. 4.
GATP	CATP			CATP	Signature:	
95-07-01 04/12/1995	TO PREVENT ENGINE FAILURE DUE TO CONNECTING ROD BOLT FAILURE, WHICH COULD RESU			Once		1. 2. 3. 4.
CATP	GATP			CATP	Signature:	
96-09-10 C 07/15/1996	TO PREVENT OIL PUMP FAILURE DUE TO IMPELLER FAILURE, WHICH COULD RESULT IN AN			Once		1. 2. 3. 4.
GATP	GATP			CATP	Signature:	
96-23-03 12/17/1996	TO PREVENT AN INFLIGHT ENGINE FAILURE DUE TO FUEL STARVATION, WHICH COULD RESU			Once		1. 2. 3. 4.
GATP	GATP			+	Signature:	
97-01-03 01/21/1997	Superseded by 97-15-11			Once		1. 2. 3. 4.
CATP	CATP			-	Signature:	
97-15-11 08/12/1997	TO PREVENT PISTON PIN FAILURE, WHICH COULD RESULT IN ENGINE FAILURE			Once		1. 2. 3. 4.
	CATP			CATP	Signature:	

Report Produced By:	Charles King	Address:			Phone:	
USARL Research Date:	Airworthiness 03/03/2004	Directive	Compliance Record		File ID	NEW
Manufacturer Textron Lycoming	<u>M</u> odel IO-540-G1D5				_	: Not Specified
AD Number Effective date	Description	Complied Date Time	Status Method of Compliance/Applicability	Once or Recur	Next Due Date Time	1. Facility 3. Cert. Num. 2. Cert. Type 4. Author. by
98-17-11 C 10/19/1998	TO PREVENT CRANKSHAFT FAILURE DUE TO CRACKING, WHICH COULD RESULT IN AN INFLIG			Once		1. 2. 3. 4.
©ATP	©ATP			©ATP	Signature:	
98-18-12 09/28/1998	Superseded by 2003-14-03			Recur		1. 2. 3. 4.
©ATP	©ATP			©ATP	Signature:	
2002-19-03 09/20/2002	To prevent crankshaft failure, which could result in total engine power loss,			Once		1. 2. 3. 4.
©ATP	©ATP			©ATP	Signature:	
2002-20-51 E 10/01/2002	Superseded by 2002-23-06			Once		1. 2. 3. 4.
©ATP	©ATP			©ATP	Signature:	
2002-23-06 11/19/2002	To prevent loss of all engine power and possible forced landing			Once		1. 2. 3. 4.
CATP	©ATP			©ATP	Signature:	
2002-26-01 01/31/2003	To prevent failure of the fuel injector fuel lines allowing fuel to spray into			Recur		1. 2. 3. 4.
©ATP	©ATP			©ATP	Signature:	

Report Produced By:	Charles King	Address:			Phone:		
JSARL Research Date:	Airworthiness	Directive	Compliance Record			NEW	
Manufacturer	03/03/2004 <u>M</u> odel				Part # :	Not Specified	
Textron Lycoming	IO-540-G1D5				Serial #: Not Specified		
AD Number Effective date	Description	Complied Date Time	Status  Method of Compliance/Applicability	Once or Recur	Next Due Date Time	1. Facility 3. Cert. Num. 2. Cert. Type 4. Author. by	
2003-14-03 08/14/2003	To prevent rotary fuel pump leaks, which could result in an engine failure, en			Recur		2. 3. 4.	
©ATP	©ATP			©ATP	Signature:		
					Signature:	1. 2. 3. 4.	
					Signature.	1.	
						2. 3. 4.	
					Signature:		
						1. 2. 3. 4.	
					Signature		
					Signature	1. 2. 3. 4.	
					Signature	1.	
						2. 3. 4.	
		STARTE E	The state of the s		Signatur	e:	

		N: 2087P	
Name:	-AR 823	N: 2087P Ser.No.: 36	
AD NO. & Rev. Date Number	Subject	Date & Hours   Method of   Compliance   Com	One- Rec- Next Comp  Auth. Sig.  time  urring Due Date  and
	SKI N 1027 - BOS	52524 S6LN 1209-	B 002034
<b>Magnetos</b> 73-07-04	Superseded by 94-01-03 R2		
74-26-09		W/A BY S/N	
78-09-07 R3	Superseded by 96-12-07	1	
78-18-04	111111111111111111111111111111111111111	WA BY MODEL	
79-12-07		W/ABY RODEL	X
79-18-06		W/A BY MAPEL	
80-17-14		W/ABY MUDES	
81-12-06 R1		W/A BY MARL	X
82-11-05		IN/A BY TWOEL	X
82-13-01		IN/ADDISITE BLOCK	ord IX I
82-20-01		WARD CI	
94-01-03 R2		IN/A BY SIN	
94-06-09	Capacitors	IN/A BY SYN	
Control of the Control		and the same of th	
96-12-07	Magneto failure	DIVE AT SOO	HRS   IX   SOO HRS
96-12-07	Magneto failure	DUE AT SOO	HRS   IX  SOOHRS
HC-92WK	P16EA	DIVE AT SOO	HRS    X  SOOHRS
96-12-07 HC-92WK 73-02-01	P16EA  Superseded by 95-11-08	DIVE AT SOO	
<b>HC-92WK</b> 73-02-01 95-11-08	P16EA  Superseded by 95-11-08  Blade separation	DIVE AT SOO	
HC-92WK 73-02-01 95-11-08 O-540 Series	P16EA  Superseded by 95-11-08  Blade separation		
HC-92WK 73-02-01 95-11-08 O-540 Series 63-14-03	P16EA  Superseded by 95-11-08  Blade separation	1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1	
HC-92WK 73-02-01 95-11-08 IO-540 Series 63-14-03 64-16-05	P16EA  Superseded by 95-11-08  Blade separation	I I I I I I I I I I I I I I I I I I I	
HC-92WK 73-02-01 95-11-08 0-540 Series 53-14-03 54-16-05 56-20-04	P16EA  Superseded by 95-11-08  Blade separation	I I I I I I I I I I I I I I I I I I I	
HC-92WK 73-02-01 95-11-08 0-540 Series 53-14-03 54-16-05 56-20-04 57-22-06	P16EA  Superseded by 95-11-08  Blade separation	INA BY SIND WABY DIM WA NO OUL FILTER WA BY MORE	
HC-92WK 73-02-01 95-11-08 0-540 Series 63-14-03 64-16-05 66-20-04 67-22-06 69-08-09	P16EA  Superseded by 95-11-08  Blade separation	INA BY SIN WABY DIM INA NO OCK FILTER INA BY MODEL INA BY MODEL	
HC-92WK 73-02-01 95-11-08 0-540 Series 53-14-03 54-16-05 56-20-04 57-22-06	P16EA  Superseded by 95-11-08  Blade separation	INA BY SAU WABY DAM INA NO OUL FILTER INA BY TOBEL INA BY TABLE WABY TABLE	
HC-92WK 73-02-01 95-11-08 0-540 Series 63-14-03 64-16-05 66-20-04 67-22-06 69-08-09	P16EA  Superseded by 95-11-08  Blade separation	INA BY SAU  NA BY SAU  NA BY FAM  NA BY TOBEL  WA BY TOBEL  WA BY TOBEL	
HC-92WK 73-02-01 95-11-08 O-540 Series 63-14-03 64-16-05 66-20-04 67-22-06 69-08-09 73-23-01	P16EA  Superseded by 95-11-08  Blade separation	I I I I I I I I I I I I I I I I I I I	
HC-92WK 73-02-01 95-11-08 0-540 Series 63-14-03 64-16-05 66-20-04 67-22-06 69-08-09 73-23-01 75-08-09	P16EA  Superseded by 95-11-08  Blade separation	I I I I I I I I I I I I I I I I I I I	
HC-92WK 73-02-01 95-11-08 O-540 Series 63-14-03 64-16-05 66-20-04 67-22-06 69-08-09 73-23-01 75-08-09	P16EA  Superseded by 95-11-08  Blade separation	I I I I I I I I I I I I I I I I I I I	
HC-92WK 73-02-01 95-11-08 0-540 Series 53-14-03 54-16-05 56-20-04 57-22-06 59-08-09 73-23-01 75-08-09 75-09-15 78-23-08	P16EA  Superseded by 95-11-08  Blade separation	I I I I I I I I I I I I I I I I I I I	

	AIRWORTHINES	SS DIRECTIVE COMPLIANCE LIST	
Name:	AR 823	N: 2087P Scr.No.: 36	
AD NO. & Rev. Date Number	Subject	Date & Hours   Method of	One- Rec- Next Comp  Auth. Sig.  time  urring Due Date  and
87-10-06 R1		IN/A BY IS/N	1 1 1 1
92-12-05	Piston pin failure	IN/A BY DATE	
92-20-07	Superseded by 93-05-21	IMPASY DATE	
93-05-21	Superseded by 93-11-11		
3-11-11	Disruption of fuel flow	INA BY HATE	FEET
5-07-01	Connecting rod bolt failure	IN/A BY DATE	
6-09-10	Oil pump failure	IN/A BY MODEL	
6-23-03	Fuel starvation	IN/A BY DIATE	
7-01-03	Superseded by 97-15-11		
7-01-04	Cylinder head separation	IN/A BY P/W	X
7-15-11	Piston pin failure	W/A BYBATE	111111
8-17-11	Crankshaft cracks	IN/A BY MODEL	
8-18-12	Rotary fuel pump leaks	N/A BYPW	